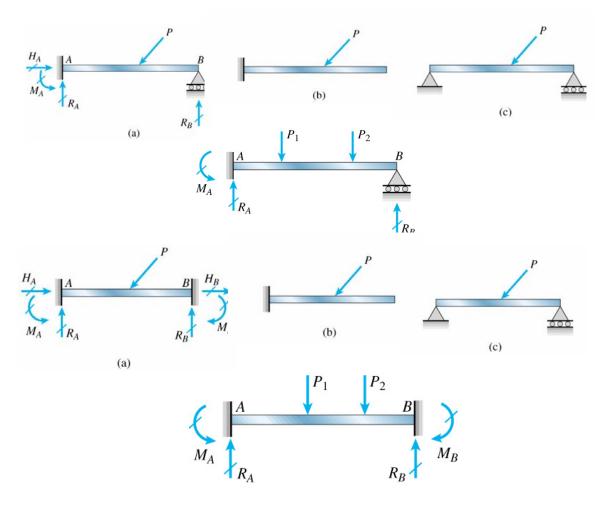
# Chapter 10 Statically Indeterminate Beams

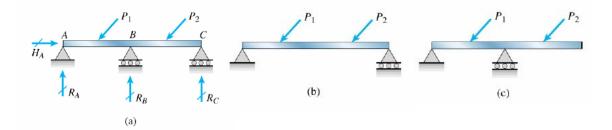
### **10.1 Introduction**

in this chapter we will analyze the beam in which the number of reactions exceed the number of independent equations of equilibrium integration of the differential equation, method of superposition compatibility equation (consistence of deformation)

### **10.2 Types of Statically Indeterminate Beams**

the number of reactions in excess of the number of equilibrium equations is called the degree of static indeterminacy





the excess reactions are called static redundants

the structure that remains when the redundants are released is called released structure or the primary structure

#### 10.3 Analysis by the Differential Equations of the Deflection Curve

$$EIv'' = M$$
  $EIv''' = V$   $EIv^{1v} = -q$ 

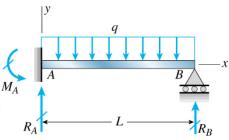
the procedure is essentially the same as that for a statically determine beam and consists of writing the differential equation, integrating to obtain its general solution, and then applying boundary and other conditions to evaluate the unknown quantities, the unknowns consist of the redundant reactions as well as the constants of integration

this method have the computational difficulties that arise when a large number of constants to be evaluated, it is practical only for relatively simple case

Example 10-1

a propped cantilever beam ABsupports a uniform load q

determine the reactions, shear forces bending moments, slopes, and deflections



choose  $R_B$  as the redundant, then

$$R_A = qL - R_B \qquad M_A = \frac{qL^2}{2} - R_B L$$

and the bending moment of the beam is

$$M = R_{A}x - M_{A} - \frac{qx^{2}}{2}$$

$$= qLx - R_{B}x - \frac{qL^{2}}{2} - R_{B}L - \frac{qx^{2}}{2}$$

$$EIv'' = M = qLx - R_{B}x - \frac{qL^{2}}{2} - R_{B}L - \frac{qx^{2}}{2}$$

$$EIv'' = \frac{qLx^{2}}{2} - \frac{R_{B}x^{2}}{2} - \frac{qL^{2}x}{2} - R_{B}Lx - \frac{qx^{3}}{6} + C_{1}$$

$$EIv = \frac{qLx^{3}}{6} - \frac{R_{b}x^{3}}{6} - \frac{qL^{2}x^{2}}{4} - \frac{R_{b}Lx^{2}}{2} - \frac{qx^{4}}{24} + C_{1}x + C_{2}$$

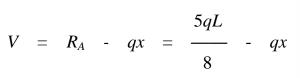
boundary conditions

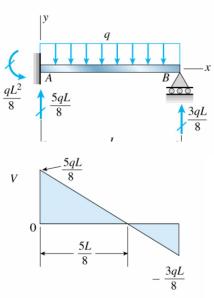
$$v(0) = 0$$
  $v'(0) = 0$   $v(L) = 0$ 

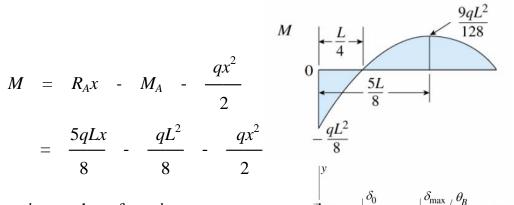
it is obtained

and 
$$C_1 = C_2 = 0$$
  $R_B = 3qL/8$   
 $R_A = 5qL/8$   
 $M_A = qL^2/8$ 

the shear force and bending moment are







 $\frac{L}{4}$ 

 $x_1$ 

the maximum shear force is

 $V_{max} = 5qL/8$  at the fixed end

the maximum positive and negative moments are

$$M_{pos} = 9qL^2/128$$
  $M_{neg} = -qL^2/8$ 

slope and deflection of the beam

$$v' = \frac{qx}{48EI} (-6L^2 + 15Lx - 8x^2)$$
$$v = -\frac{qx^2}{48EI} (3L^2 - 5Lx + 2x^2)$$

to determine the  $\delta_{max}$ , set v' = 0

$$-6L^2$$
 + 15Lx -  $8x^2$  = 0

we have  $x_1 = 0.5785L$ 

$$\delta_{max} = -v(x_1) = 0.005416 \frac{qL^4}{EI}$$

the point of inflection is located at M = 0, i.e. x = L/4

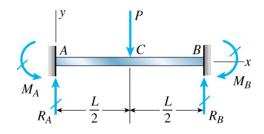
 $\kappa$  < 0 and M < 0 for x < L/4 $\kappa$  > 0 and M > 0 for x > L/4 the slope at B is

$$\theta_B = (y')_{x=L} = \frac{qL^3}{48EI}$$

Example 10-2

a fixed-end beam ABC supports a concentrated load P at the midpoint

determine the reactions, shear forces, bending moments, slopes, and deflections



because the load P in vertical direction and symmetric

$$H_A = H_B = 0 \qquad R_A = R_B = P/2$$

$$M_A = M_B$$
 (1 degree of indeterminacy)

$$M = \frac{Px}{2} - M_A \quad (0 \le x \le L/2)$$
$$EIv'' = M = \frac{Px}{2} - M_A \quad (0 \le x \le L/2)$$

after integration, it is obtained

$$EIv' = \frac{Px^2}{4} - M_A x + C_1 \quad (0 \le x \le L/2)$$
$$EIv = \frac{Px^3}{12} - \frac{M_A x^2}{2} + C_1 x + C_2 \quad (0 \le x \le L/2)$$

boundary conditions

v(0) = 0 v'(0) = 0

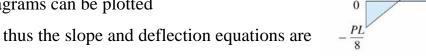
symmetric condition

$$v'(0) = 0$$

the constants  $C_1$ ,  $C_2$  and the moment  $M_A$  are obtained

$$C_1 = C_2 = 0$$
$$M_A = \frac{PL}{8} = M_B$$

the shear force and bending moment diagrams can be plotted



$$v' = -\frac{Px}{8EI}(L - 2x) \qquad (0 \le x \le L/2)$$
$$v = -\frac{Px^2}{48EI}(3L - 4x) \qquad (0 \le x \le L/2)$$

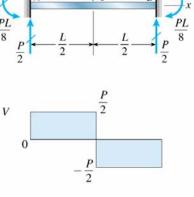
the maximum deflection occurs at the center

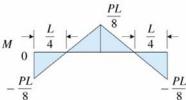
$$\delta_{max} = -v(L/2) = \frac{PL^3}{192EI}$$

the point of inflection occurs at the point where M = 0, i.e. x = L/4, the deflection at this point is

$$\delta = -v(L/4) = \frac{PL^3}{384EI}$$
which is equal  $\delta_{max}/2$ 

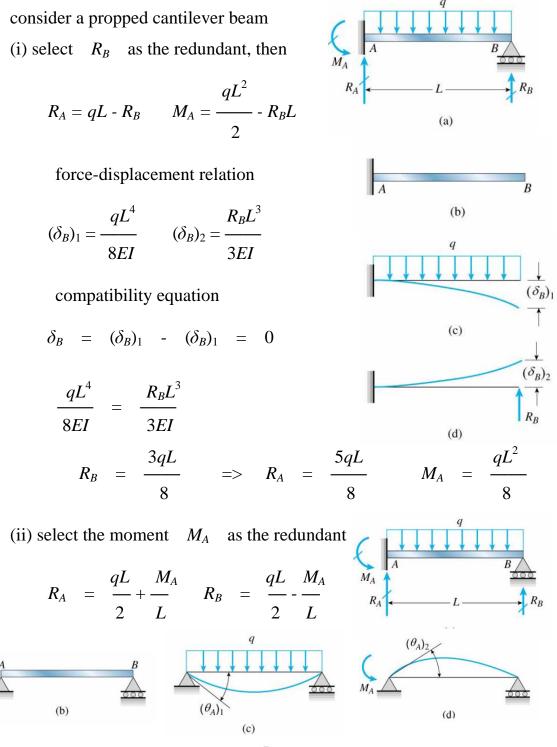
$$A \downarrow 0 C \downarrow 0_{max} \downarrow 0 B -x$$





#### **10.4 Method of Superposition**

- 1. selecting the reaction redundants
- 2. establish the force-displacement relations
- 3. consistence of deformation (compatibility equation)



force-displacement relation

$$(\theta_A)_1 = \frac{qL^3}{24EI} (\theta_A)_2 = \frac{M_A L}{3EI}$$

compatibility equation

$$\theta_A = (\theta_A)_1 - (\theta_A)_2 = \frac{qL^3}{24EI} - \frac{M_AL}{3EI} = 0$$

thus  $M_A = qL^2/8$ and  $R_A = 5qL/8$   $R_B = 3qL/8$ 

Example 10-3

a continuous beam ABC supports a uniform load q

determine the reactions

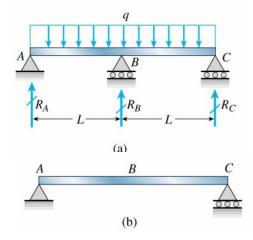
select  $R_B$  as the redundant, then

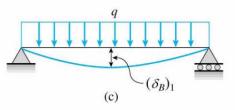
$$R_A = R_C = qL - \frac{qL}{2}$$

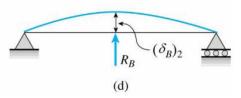
force-displacement relation

$$(\delta_B)_1 = \frac{5qL(2L)^4}{384EI} = \frac{5qL^4}{24EI}$$
$$(\delta_B)_2 = \frac{R_B(2L)^3}{48EI} = \frac{R_BL^3}{6EI}$$

compatibility equation







$$\delta_B = (\delta_B)_1 - (\delta_B)_2 = \frac{5qL^4}{24EI} - \frac{R_BL^3}{6EI} = 0$$

thus  $R_B = 5qL/4$ and  $R_A = R_C = 3qL/8$ 

Example 10-4

a fixed-end beam *AB* is loaded by a force *P* acting at point *D* determine reactions at the ends also determine  $\delta_D$ 

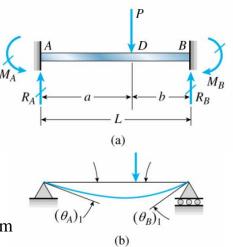
this is a 2-degree of indeterminacy problem select  $M_A$  and  $M_B$  as the redundants

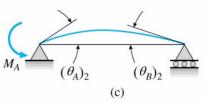
$$R_A = \frac{Pb}{L} + \frac{M_A}{L} - \frac{M_B}{L}$$
$$R_B = \frac{Pa}{L} - \frac{M_A}{L} + \frac{M_B}{L}$$

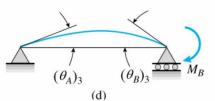
force-displacement relations

 $(\theta_A)_1 = \frac{Pab(L+b)}{6LEI} \qquad (\theta_B)_1 = \frac{Pab(L+a)}{6LEI}$  $(\theta_A)_2 = \frac{M_A L}{3EI} (\theta_B)_2 = \frac{M_A L}{6EI}$  $(\theta_A)_3 = \frac{M_B L}{6EI} (\theta_B)_3 = \frac{M_B L}{3EI}$ 

compatibility equations







$$\theta_A = (\theta_A)_1 - (\theta_A)_2 - (\theta_A)_3 = 0$$
  
$$\theta_B = (\theta_B)_1 - (\theta_B)_2 - (\theta_B)_3 = 0$$

i.e. 
$$\frac{M_AL}{3EI} + \frac{M_BL}{6EI} = \frac{Pab(L+b)}{6LEI}$$
  
 $\frac{M_AL}{6EI} + \frac{M_BL}{3EI} = \frac{Pab(L+a)}{6LEI}$ 

solving these equations, we obtain

$$M_A = \frac{Pab^2}{L^2}$$
  $M_B = \frac{Pa^2b}{L^2}$ 

and the reactions are

$$R_A = \frac{Pb^2}{L^3}(L+2a)$$
  $R_B = \frac{Pa^2}{L^3}(L+2b)$ 

the deflection  $\delta_D$  can be expressed as

$$\delta_{D} = (\delta_{D})_{1} - (\delta_{D})_{2} - (\delta_{D})_{3}$$

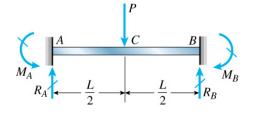
$$(\delta_{D})_{1} = \frac{Pa^{2}b^{2}}{3LEI}$$

$$(\delta_{D})_{2} = \frac{M_{A}ab}{6LEI}(L + b) = \frac{Pa^{2}b^{3}}{6L^{3}EI}(L + b)$$

$$(\delta_{D})_{3} = \frac{M_{B}ab}{6LEI}(L + a) = \frac{Pa^{3}b^{2}}{6L^{3}EI}(L + a)$$

$$Pa^{3}b^{3}$$

thus  $\delta_D = \frac{Pa v}{3L^3 EI}$ if a = b = L/2



then 
$$M_A = M_B = \frac{PL}{8}$$
  $R_A = R_B = \frac{P}{2}$   
and  $\delta_C = \frac{PL^3}{192EI}$ 

Example 10-5

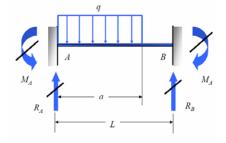
a fixed-end beam AB supports a uniform load q acting over part of the span

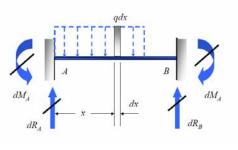
determine the reactions of the beam

to obtain the moments caused by qdx, replace P to qdx, a to x, and b to L - x

$$dM_A = \frac{qx(L-x)^2 dx}{L^2}$$
$$dM_B = \frac{qx^2(L-x) dx}{L^2}$$

integrating over the loaded part





$$M_A = \int dM_A = \frac{q}{L^2} \int_0^a x(L - x)^2 dx = \frac{qa^2}{12L^2} (6L^2 - 8aL + 3a^2)$$
$$M_B = \int dM_B = \frac{q}{L^2} \int_0^a x^2(L - x) dx = \frac{qa^3}{12L^2} (4L^2 - 3a)$$

Similarly

$$dR_A = \frac{q(L-x)^2(L+2x)dx}{L^3}$$
$$dR_B = \frac{qx^2(3L-2x)dx}{L^3}$$

.

integrating over the loaded part

$$R_{A} = \int dR_{A} = \frac{q}{L^{3}} \int_{0}^{a} (L - x)^{2} (L + 2x) dx = \frac{qa}{2L^{3}} (2L^{3} - 2a^{2}L + a^{3})$$
$$R_{B} = \int dR_{B} = \frac{q}{L^{3}} \int_{0}^{a} x^{2} (3L - 2x) dx = \frac{qa^{3}}{2L^{3}} (2L - a)$$

for the uniform acting over the entire length, i.e. a = L

$$M_A = M_B = \frac{qL^2}{12}$$

$$R_A = R_B = \frac{qL}{2}$$

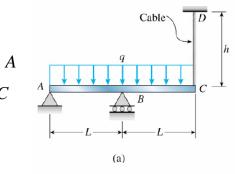
$$R_A = R_B = \frac{qL}{2}$$

the center point deflections due to uniform load and the end moments are

$$(\delta_C)_1 = \frac{5qL^4}{384EI} \qquad (\delta_C)_2 = \frac{M_A L}{8EI} = \frac{(qL^2/12)L^2}{8EI} = \frac{qL^4}{96EI}$$
$$\delta_C = (\delta_C)_1 - (\delta_C)_2 = \frac{qL^4}{384EI}$$

Example 10-6

a beam ABC rests on supports Aand B and is supported by a cable at C

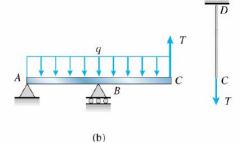


find the force T of the cable

take the cable force T as redundant

the deflection  $(\delta_C)_1$  due the uniform load can be found from example 9.9 with a = L

$$(\delta_C)_1 = \frac{qL^4}{4E_bI_b}$$



the deflection  $(\delta_C)_2$  due to a force T acting on C is obtained

use conjugate beam method

$$(\delta_C)_2 = M = \frac{TL^2}{3E_b I_b}L + \frac{TL}{E_b I_b}\frac{L}{2}\frac{2L}{3}$$
$$= \frac{2TL^3}{3E_b I_b}$$

the elongation of the cable is

$$(\delta_C)_3 = \frac{Th}{E_c A_c}$$

compatibility equation

$$(\delta_C)_1 - (\delta_C)_2 = (\delta_C)_3$$

$$\frac{qL^4}{4E_bI_b} - \frac{2TL^3}{3E_bI_b} = \frac{Th}{E_cA_c}$$

$$T = \frac{3qL^4E_cA_c}{8L^3E_cA_c + 12hE_bI_b}$$

**10.5 Temperature Effects** 

# **10.6 Longitudinal Displacements at the Ends of the Beams**