

# Chapter 3 Torsion

## 3.1 Introduction

Torsion : twisting of a structural member, when it is loaded by couples that produce rotation about its longitudinal axis

$$T_1 = P_1 d_1 \quad T_2 = P_2 d_2$$

the couples  $T_1$ ,  $T_2$  are called torques, twisting couples or twisting moments

unit of  $T$  : N-m, lb-ft

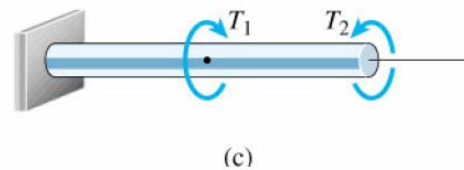
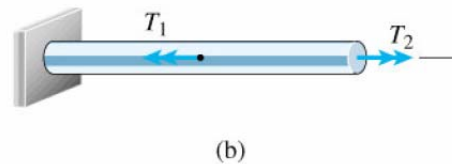
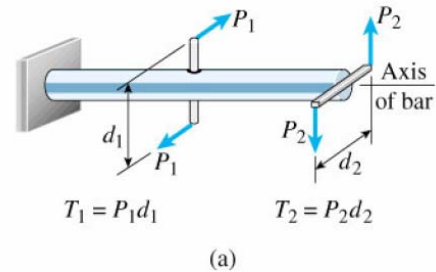
in this chapter, we will develop formulas for the stresses and deformations produced in circular bars subjected to torsion, such as drive shafts, thin-walled members

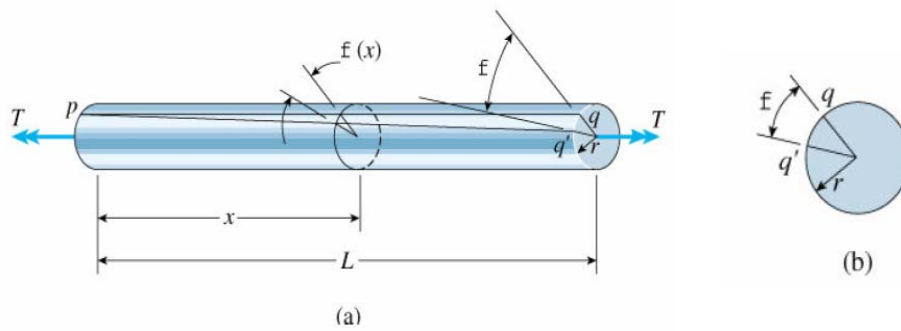
analysis of more complicated shapes required more advanced method than those presented here

this chapter cover several additional topics related to torsion, such statically indeterminate members, strain energy, thin-walled tube of noncircular section, stress concentration, and nonlinear behavior

## 3.2 Torsional Deformation of a Circular Bar

consider a bar or shaft of circular cross section twisted by a couple  $T$ , assume the left-hand end is fixed and the right-hand end will rotate a small angle  $\phi$ , called angle of twist



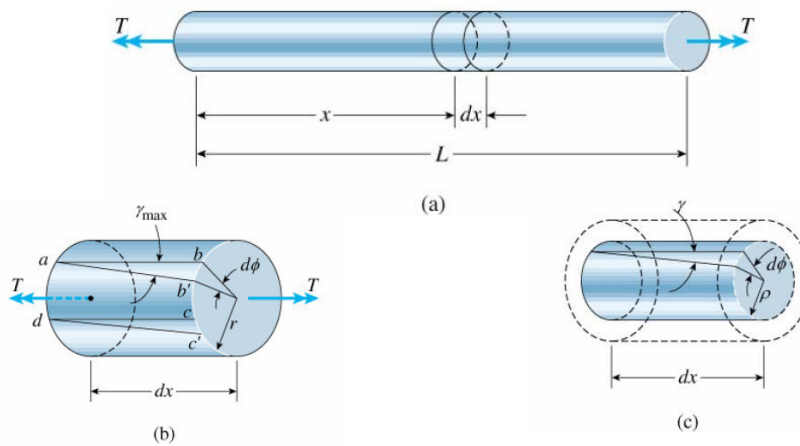


if every cross section has the same radius and subjected to the same torque, the angle  $\phi(x)$  will vary linearly between ends

under twisting deformation, it is assumed

1. plane section remains plane
2. radii remaining straight and the cross sections remaining plane and circular
3. if  $\phi$  is small, neither the length  $L$  nor its radius will change

consider an element of the bar  $dx$ , on its outer surface we choose an small element  $abcd$ ,



during twisting the element rotate a small angle  $d\phi$ , the element is in a state of pure shear, and deformed into  $ab'c'd$ , its shear strain  $\gamma_{\max}$  is

$$\gamma_{\max} = \frac{b b'}{a b} = \frac{r d\phi}{dx}$$

$d\phi / dx$  represents the rate of change of the angle of twist  $\phi$ , denote  $\theta = d\phi / dx$  as the angle of twist per unit length or the rate of twist, then

$$\gamma_{\max} = r \theta$$

in general,  $\phi$  and  $\theta$  are function of  $x$ , in the special case of pure torsion,  $\theta$  is constant along the length (every cross section is subjected to the same torque)

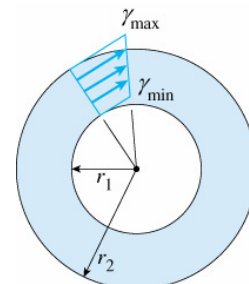
$$\theta = \frac{\phi}{L} \quad \text{then} \quad \gamma_{\max} = \frac{r \phi}{L}$$

and the shear strain inside the bar can be obtained

$$\gamma = \rho \theta = \frac{\rho}{r} \gamma_{\max}$$

for a circular tube, it can be obtained

$$\gamma_{\min} = \frac{r_1}{r_2} \gamma_{\max}$$



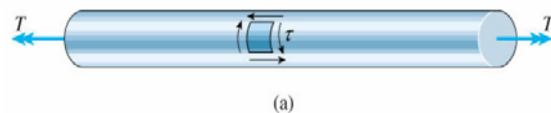
the above relationships are based only upon geometric concepts, they are valid for a circular bar of any material, elastic or inelastic, linear or nonlinear

### 3.3 Circular Bars of Linearly Elastic Materials

shear stress  $\tau$  in the bar of a linear elastic material is

$$\tau = G \gamma$$

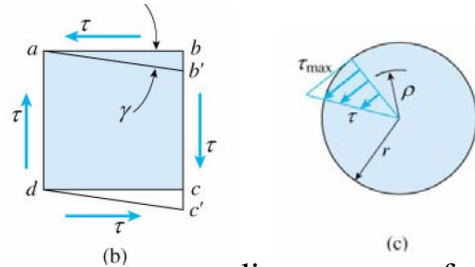
$G$  : shear modulus of elasticity



with the geometric relation of the shear strain, it is obtained

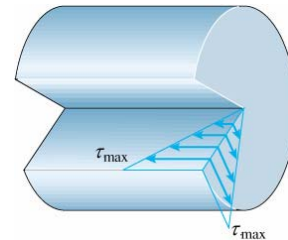
$$\tau_{\max} = G r \theta$$

$$\tau = G \rho \theta = \frac{\rho}{r} \tau_{\max}$$



$\tau$  and  $\gamma$  in circular bar vary linear with the radial distance  $\rho$  from the center, the maximum values  $\tau_{\max}$  and  $\gamma_{\max}$  occur at the outer surface

the shear stress acting on the plane of the cross section are accompanied by shear stresses of the same magnitude acting on longitudinal plane of the bar



if the material is weaker in shear on longitudinal plane than on cross-sectional

planes, as in the case of a circular bar made of wood, the first crack due to twisting will appear on the surface in longitudinal direction

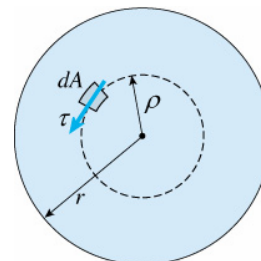
a rectangular element with sides at  $45^\circ$  to the axis of the shaft will be subjected to tensile and compressive stresses



### The Torsion Formula

consider a bar subjected to pure torsion, the shear force acting on an element  $dA$  is  $\tau dA$ , the moment of this force about the axis of bar is  $\tau \rho dA$

$$dM = \tau \rho dA$$



equation of moment equilibrium

$$\begin{aligned} T &= \int_A dM = \int_A \tau \rho dA = \int_A G \theta \rho^2 dA = G \theta \int_A \rho^2 dA \\ &= G \theta I_p \quad [\tau = G \theta \rho] \end{aligned}$$

in which  $I_p = \int_A \rho^2 dA$  is the polar moment of inertia

$$I_p = \frac{\pi r^4}{2} = \frac{\pi d^4}{32} \quad \text{for circular cross section}$$

the above relation can be written

$$\theta = \frac{T}{G I_p}$$

$G I_p$  : torsional rigidity

the angle of twist  $\phi$  can be expressed as

$$\phi = \theta L = \frac{TL}{G I_p} \quad \phi \text{ is measured in radians}$$

torsional flexibility  $f = \frac{L}{G I_p}$

torsional stiffness  $k = \frac{G I_p}{L}$

and the shear stress is

$$\tau = G \rho \theta = G \rho \frac{T}{G I_p} = \frac{T \rho}{I_p}$$

the maximum shear stress  $\tau_{\max}$  at  $\rho = r$  is

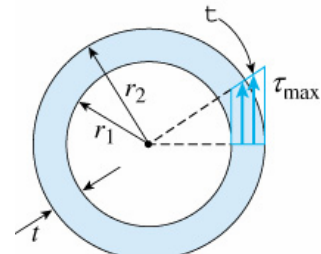
$$\tau_{\max} = \frac{T r}{I_p} = \frac{16 T}{\pi d^3}$$

for a circular tube

$$I_p = \pi (r_2^4 - r_1^4) / 2 = \pi (d_2^4 - d_1^4) / 32$$

if the hollow tube is very thin

$$\begin{aligned} I_p &\simeq \pi (r_2^2 + r_1^2) (r_2 + r_1) (r_2 - r_1) / 2 \\ &= \pi (2r^2) (2r) (t) = 2 \pi r^3 t = \pi d^3 t / 4 \end{aligned}$$



limitations

1. bar have circular cross section (either solid or hollow)
2. material is linear elastic

note that the above equations cannot be used for bars of noncircular shapes, because their cross sections do not remain plane and their maximum stresses are not located at the farthest distances from the midpoint

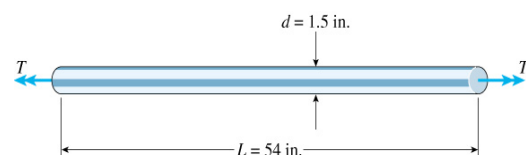
Example 3-1

a solid bar of circular cross section

$$d = 40 \text{ mm}, \quad L = 1.3 \text{ m}, \quad G = 80 \text{ GPa}$$

$$(a) \quad T = 340 \text{ N-m}, \quad \tau_{\max}, \quad \phi = ?$$

$$(b) \quad \tau_{\text{all}} = 42 \text{ MPa}, \quad \phi_{\text{all}} = 2.5^\circ, \quad T = ?$$



$$(a) \quad \tau_{\max} = \frac{16 T}{\pi d^3} = \frac{16 \times 340 \text{ N-M}}{\pi (0.04 \text{ m})^3} = 27.1 \text{ MPa}$$

$$I_p = \pi d^4 / 32 = 2.51 \times 10^{-7} \text{ m}^4$$

$$\phi = \frac{T L}{G I_p} = \frac{340 \text{ N-m} \times 1.3 \text{ m}}{80 \text{ GPa} \times 2.51 \times 10^{-7} \text{ m}^4} = 0.02198 \text{ rad} = 1.26^\circ$$

(b) due to  $\tau_{all} = 42 \text{ MPa}$

$$T_1 = \pi d^3 \tau_{all} / 16 = \pi (0.04 \text{ m})^3 \times 42 \text{ MPa} / 16 = 528 \text{ N-m}$$

$$\text{due to } \phi_{all} = 2.5^\circ = 2.5 \times \pi \text{ rad} / 180^\circ = 0.04363 \text{ rad}$$

$$\begin{aligned} T_2 &= G I_p \phi_{all} / L = 80 \text{ GPa} \times 2.51 \times 10^{-7} \text{ m}^4 \times 0.04363 / 1.3 \text{ m} \\ &= 674 \text{ N-m} \end{aligned}$$

$$\text{thus } T_{all} = \min [T_1, T_2] = 528 \text{ N-m}$$

### Example 3-2

a steel shaft of either solid bar or circular tube

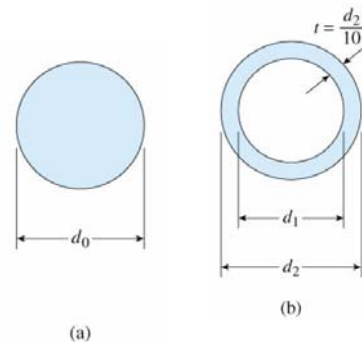
$$T = 1200 \text{ N-m}, \quad \tau_{all} = 40 \text{ MPa}$$

$$\theta_{all} = 0.75^\circ / \text{m} \quad G = 78 \text{ GPa}$$

(a) determine  $d_0$  of the solid bar

(b) for the hollow shaft,  $t = d_2 / 10$ , determine  $d_2$

(c) determine  $d_2 / d_0$ ,  $W_{hollow} / W_{solid}$



(a) for the solid shaft, due to  $\tau_{all} = 40 \text{ MPa}$

$$d_0^3 = 16 T / \pi \tau_{all} = 16 \times 1200 / \pi \times 40 = 152.8 \times 10^{-6} \text{ m}^3$$

$$d_0 = 0.0535 \text{ m} = 53.5 \text{ mm}$$

$$\text{due to } \theta_{all} = 0.75^\circ / \text{m} = 0.75 \times \pi \text{ rad} / 180^\circ / \text{m} = 0.01309 \text{ rad} / \text{m}$$

$$I_p = T / G \theta_{all} = 1200 / 78 \times 10^9 \times 0.01309 = 117.5 \times 10^{-8} \text{ m}^4$$

$$d_0^4 = 32 I_p / \pi = 32 \times 117.5 \times 10^{-8} / \pi = 1197 \times 10^{-8} \text{ m}^4$$

$$d_0 = 0.0588 \text{ m} = 58.8 \text{ mm}$$

thus, we choose  $d_0 = 58.8 \text{ mm}$  [in practical design,  $d_0 = 60 \text{ mm}$ ]

(b) for the hollow shaft

$$d_1 = d_2 - 2t = d_2 - 0.2d_2 = 0.8d_2$$

$$I_p = \pi (d_2^4 - d_1^4) / 32 = \pi [d_2^4 - (0.8d_2)^4] / 32 = 0.05796 d_2^4$$

due to  $\tau_{all} = 40 \text{ MPa}$

$$I_p = 0.05796 d_2^4 = T r / \tau_{all} = 1200 (d_2/2) / 40$$

$$d_2^3 = 258.8 \times 10^{-6} \text{ m}^3$$

$$d_2 = 0.0637 \text{ m} = 63.7 \text{ mm}$$

due to  $\theta_{all} = 0.75^\circ / \text{m} = 0.01309 \text{ rad} / \text{m}$

$$\theta_{all} = 0.01309 = T / G I_p = 1200 / 78 \times 10^9 \times 0.05796 d_2^4$$

$$d_2^4 = 2028 \times 10^{-8} \text{ m}^4$$

$$d_2 = 0.0671 \text{ m} = 67.1 \text{ mm}$$

thus, we choose  $d_0 = 67.1 \text{ mm}$  [in practical design,  $d_0 = 70 \text{ mm}$ ]

(c) the ratios of hollow and solid bar are

$$d_2 / d_0 = 67.1 / 58.8 = 1.14$$

$$\frac{W_{\text{hollow}}}{W_{\text{solid}}} = \frac{A_{\text{hollow}}}{A_{\text{solid}}} = \frac{\pi (d_2^2 - d_1^2) / 4}{\pi d_0^2 / 4} = 0.47$$

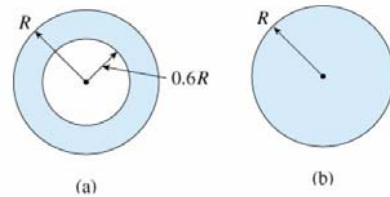
the hollow shaft has 14% greater in diameter but 53% less in weight

### Example 3-3

a hollow shaft and a solid shaft has same material, same length, same outer radius  $R$ , and  $r_i = 0.6R$  for the hollow shaft

(a) for same  $T$ , compare their  $\tau$ ,  $\theta$ , and  $W$

(b) determine the strength-to-weight ratio



$$(a) \quad \therefore \tau = T R / I_p \quad \theta = T L / G I_p$$

$\therefore$  the ratio of  $\tau$  or  $\theta$  is the ratio of  $1 / I_p$

$$(I_p)_H = \pi R^2 / 2 - \pi (0.6R)^2 / 2 = 0.4352 \pi R^2$$



$$(I_p)_S = \pi R^2 / 2 = 0.5 \pi R^2$$

$$(I_p)_S / (I_p)_H = 0.5 / 0.4352 = 1.15$$

thus  $\beta_1 = \tau_H / \tau_S = (I_p)_S / (I_p)_H = 1.15$

also  $\beta_2 = \phi_H / \phi_S = (I_p)_S / (I_p)_H = 1.15$

$$\beta_3 = W_H / W_S = A_H / A_S = \pi [R^2 - (0.6R)^2] / \pi R^2 = 0.64$$

the hollow shaft has 15% greater in  $\tau$  and  $\phi$ , but 36% decrease in weight

(b) strength-to-weight ratio  $S = T_{all} / W$

$$T_H = \tau_{max} I_p / R = \tau_{max} (0.4352 \pi R^4) / R = 0.4352 \pi R^3 \tau_{max}$$

$$T_S = \tau_{max} I_p / R = \tau_{max} (0.5 \pi R^4) / R = 0.5 \pi R^3 \tau_{max}$$

$$W_H = 0.64 \pi R^2 L \gamma \quad W_S = \pi R^2 L \gamma$$

thus  $S_H = T_H / W_H = 0.68 \tau_{max} R / \gamma L$

$$S_S = T_S / W_S = 0.5 \tau_{max} R / \gamma L$$

$S_H$  is 36% greater than  $S_S$

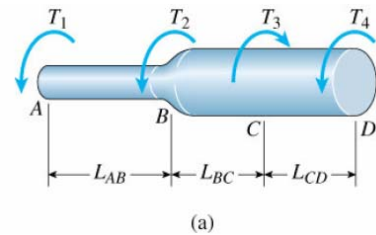
### 3.4 Nonuniform Torsion

(1) constant torque through each segment

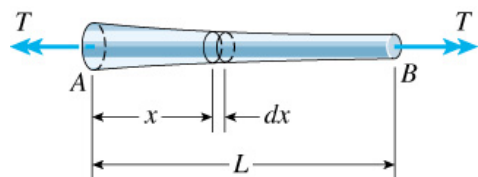
$$T_{CD} = -T_1 - T_2 + T_3$$

$$T_{BC} = -T_1 - T_2 \quad T_{AB} = -T_1$$

$$\phi = \sum_{i=1}^n \phi_i = \sum_{i=1}^n \frac{T_i L_i}{G_i I_{pi}}$$



(2) constant torque with continuously varying cross section

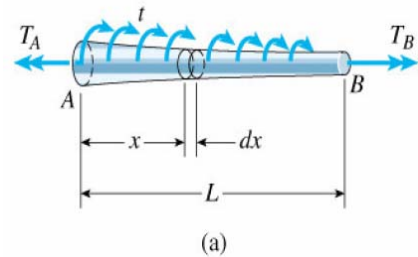


$$d\phi = \frac{T dx}{G I_p(x)}$$

$$\phi = \int_0^L d\phi = \int_0^L \frac{T dx}{G I_p(x)}$$

(3) continuously varying cross section and continuously varying torque

$$\phi = \int_0^L d\phi = \int_0^L \frac{T(x) dx}{G I_p(x)}$$



### Example 3-4

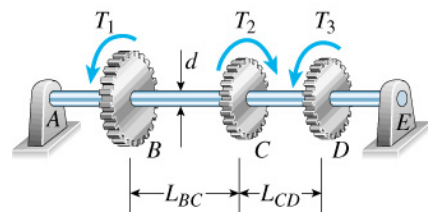
a solid steel shaft *ABCDE*,  $d = 30$  mm

$$T_1 = 275 \text{ N-m} \quad T_2 = 450 \text{ N-m}$$

$$T_3 = 175 \text{ N-m} \quad G = 80 \text{ GPa}$$

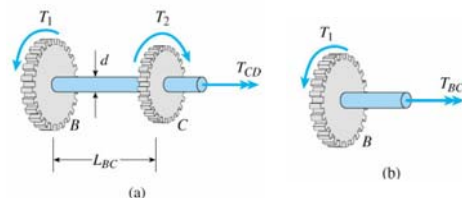
$$L_1 = 500 \text{ mm} \quad L_2 = 400 \text{ mm}$$

determine  $\tau_{max}$  in each part and  $\phi_{BD}$



$$T_{CD} = T_2 - T_1 = 175 \text{ N-m}$$

$$T_{BC} = -T_1 = -275 \text{ N-m}$$



$$\tau_{BC} = \frac{16 T_{BC}}{\pi d^3} = \frac{16 \times 275 \times 10^3}{\pi 30^3} = 51.9 \text{ MPa}$$

$$\tau_{CD} = \frac{16 T_{CD}}{\pi d^3} = \frac{16 \times 175 \times 10^3}{\pi 30^3} = 33 \text{ MPa}$$

$$\phi_{BD} = \phi_{BC} + \phi_{CD}$$

$$I_p = \frac{\pi d^4}{32} = \frac{\pi 30^4}{32} = 79,520 \text{ mm}^2$$

$$\phi_{BC} = \frac{T_{BC} L_1}{G I_p} = \frac{-275 \times 10^3 \times 500}{80 \times 10^3 \times 79,520} = -0.0216 \text{ rad}$$

$$\phi_{CD} = \frac{T_{CD} L_2}{G I_p} = \frac{175 \times 10^3 \times 400}{80 \times 10^3 \times 79,520} = 0.011 \text{ rad}$$

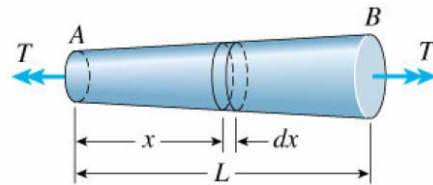
$$\phi_{BD} = \phi_{BC} + \phi_{CD} = -0.0216 + 0.011 = -0.0106 \text{ rad} = -0.61^\circ$$

### Example 3-5

a tapered bar AB of solid circular cross section is twisted by torque  $T$

$$d = d_A \text{ at } A, \quad d = d_B \text{ at } B, \quad d_B \geq d_A$$

determine  $\tau_{max}$  and  $\phi$  of the bar



(a)  $T = \text{constant over the length,}$

thus  $\tau_{max}$  occurs at  $d_{min}$  [end A]

$$\tau_{max} = \frac{16 T}{\pi d_A^3}$$



(b) angle of twist

$$d(x) = d_A + \frac{d_B - d_A}{L} x$$

$$I_p(x) = \frac{\pi d^4}{32} = \frac{\pi}{32} \left( d_A + \frac{d_B - d_A}{L} x \right)^4$$

then

$$\phi = \int_0^L \frac{T dx}{G I_p(x)} = \frac{32 T}{\pi G} \int_0^L \frac{dx}{\left( d_A + \frac{d_B - d_A}{L} x \right)^4}$$

to evaluate the integral, we note that it is of the form

$$\int \frac{dx}{(a + bx)^4} = -\frac{1}{3b(a + bx)^3}$$

if we choose  $a = d_A$  and  $b = (d_B - d_A) / L$ , then the integral of  $\phi$  can be obtained

$$\phi = \frac{32 TL}{3\pi G(d_B - d_A)} \left( \frac{1}{d_A^3} - \frac{1}{d_B^3} \right)$$

a convenient form can be written

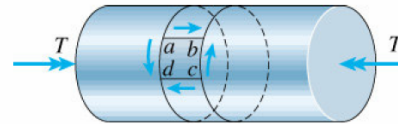
$$\phi = \frac{TL}{GI_{pA}} \left( \frac{\beta^2 + \beta + 1}{3\beta^3} \right)$$

where  $\beta = d_B / d_A$   $I_{pA} = \pi d_A^4 / 32$

in the special case of a prismatic bar,  $\beta = 1$ , then  $\phi = TL / GI_p$

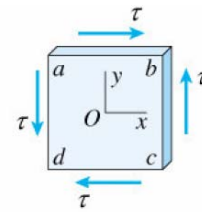
### 3.5 Stresses and Strains in Pure Shear

for a circular bar subjected to torsion, shear stresses act over the cross sections and on longitudinal planes



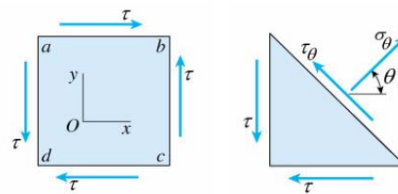
(a)

an stress element  $abcd$  is cut between two cross sections and between two longitudinal planes, this element is in a state of pure shear



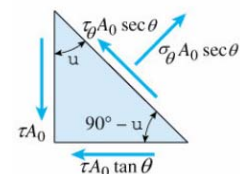
(b)

we now cut from the plane stress element to a wedge-shaped element, denote  $A_0$  the area of the vertical side face, then the area of the bottom face is  $A_0 \tan \theta$ ,



(a)

(b)



(c)

and the area of the inclined face is  $A_0 \sec \theta$

summing forces in the direction of  $\sigma_\theta$

$$\sigma_\theta A_0 \sec \theta = \tau A_0 \sin \theta + \tau A_0 \tan \theta \cos \theta$$

or 
$$\sigma_\theta = 2 \tau \sin \theta \cos \theta = \tau \sin 2\theta$$

summing forces in the direction of  $\tau_\theta$

$$\tau_\theta A_0 \sec \theta = \tau A_0 \cos \theta - \tau A_0 \tan \theta \sin \theta$$

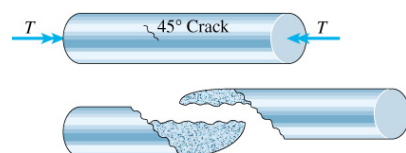
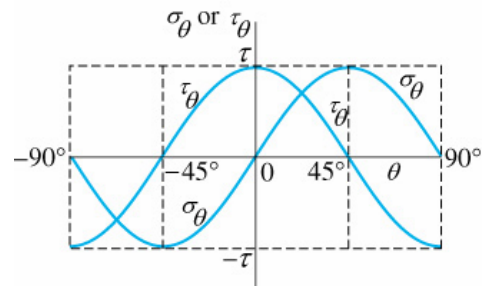
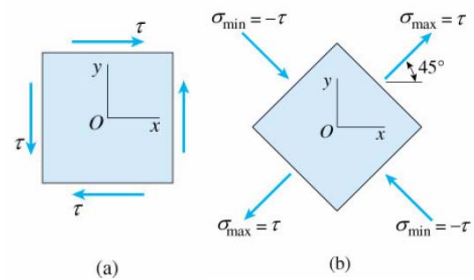
or 
$$\tau_\theta = \tau (\cos^2 \theta - \sin^2 \theta) = \tau \cos 2\theta$$

$\sigma_\theta$  and  $\tau_\theta$  vary with  $\theta$  is plotted in figure

$$\begin{aligned} (\tau_\theta)_{max} &= \tau & \text{at } \theta &= 0^\circ \\ (\tau_\theta)_{min} &= -\tau & \text{at } \theta &= \pm 90^\circ \\ (\sigma_\theta)_{max} &= \pm \tau & \text{at } \theta &= \pm 45^\circ \end{aligned}$$

the state of pure shear stress is equivalent to equal tensile and compressive stresses on an element rotation through an angle of  $45^\circ$

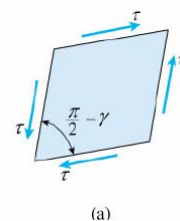
if a twisted bar is made of material that is weaker in tension than in shear, failure will occur in tension along a helix inclined at  $45^\circ$ , such as chalk



Strains in pure shear

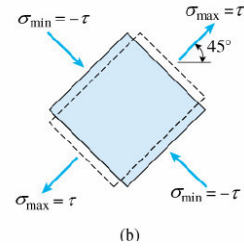
if the material is linearly elastic

$$\gamma = \tau / G$$



where  $G$  is the shear modulus of elasticity

consider the strains that occur in an element oriented at  $\theta = 45^\circ$ ,  $\sigma_{max} = \tau$  applied at  $45^\circ$  and  $\sigma_{min} = -\tau$  applied at  $\theta = -45^\circ$



then at  $\theta = 45^\circ$

$$\epsilon_{max} = \frac{\sigma_{max}}{E} - \frac{\nu \sigma_{min}}{E} = \frac{\tau}{E} + \frac{\nu \tau}{E} = \frac{\tau}{E} (1 + \nu)$$

at  $\theta = -45^\circ$   $\epsilon = -\epsilon_{max} = -\tau (1 + \nu) / E$

it will be shown in next section the following relationship

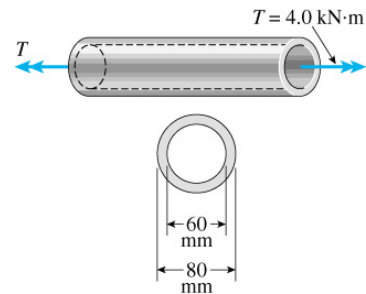
$$\epsilon_{max} = \frac{\gamma}{2}$$

### Example 3-6

a circular tube with  $d_o = 80$  mm,  $d_i = 60$  mm

$T = 4$  kN-m  $G = 27$  GPa

determine (a) maximum tensile, compressive and shear stresses (b) maximum strains



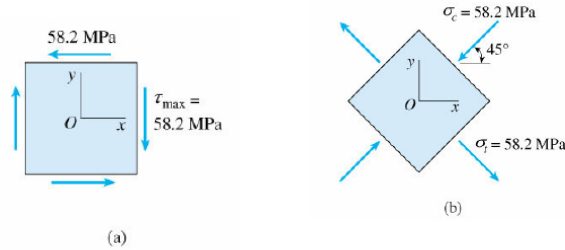
(a) the maximum shear stress is

$$\tau_{max} = \frac{T r}{I_p} = \frac{4000 \times 0.04}{\frac{\pi}{32} [(0.08)^4 - (0.06)^4]} = 58.2 \text{ MPa}$$

the maximum tensile and compressive stresses are

$$\sigma_t = 58.2 \text{ MPa} \quad \text{at } \theta = -45^\circ$$

$$\sigma_c = -58.2 \text{ MPa} \quad \text{at } \theta = 45^\circ$$



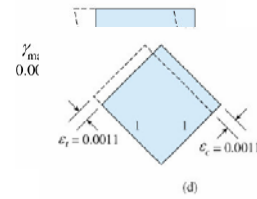
(b) maximum strains

$$\gamma_{max} = \tau_{max} / G = 58.2 / 27 \times 10^3 = 0.0022$$

the maximum normal strains is

$$\epsilon_{max} = \gamma_{max} / 2 = 0.011$$

i.e.  $\epsilon_t = 0.011$        $\epsilon_c = -0.011$



### 3.6 Relationship Between Moduli of Elasticity $E$ , $G$ and $\nu$

an important relationship between  $E$ ,  $G$  and  $\nu$  can be obtained

consider the square stress element  $abcd$ , with the length of each side denoted as  $h$ , subjected to pure shear stress  $\tau$ , then

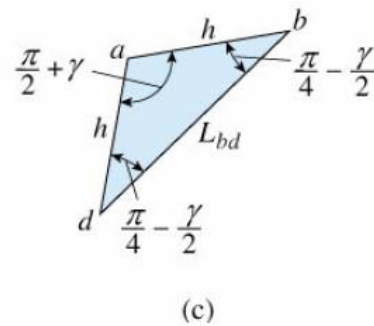
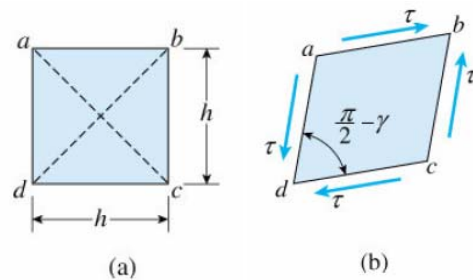
$$\gamma = \tau / G$$

the length of diagonal  $bd$  is  $\sqrt{2} h$ , after deformation

$$L_{bd} = \sqrt{2} h (1 + \epsilon_{max})$$

using the law of cosines for  $\triangle abd$

$$L_{bd}^2 = h^2 + h^2 - 2 h^2 \cos \left( \frac{\pi}{2} + \gamma \right) = 2 h^2 \left[ 1 - \cos \left( \frac{\pi}{2} + \gamma \right) \right]$$



then  $(1 + \varepsilon_{max})^2 = 1 - \cos\left(\frac{\pi}{2} + \gamma\right) = 1 + \sin \gamma$

thus  $1 + 2\varepsilon_{max} + \varepsilon_{max}^2 = 1 + \sin \gamma$

$\therefore \varepsilon_{max}$  is very small, then  $\varepsilon_{max}^2 \rightarrow 0$ , and  $\sin \gamma \rightarrow \gamma$

the resulting expression can be obtained

$$\varepsilon_{max} = \gamma / 2$$

with  $\varepsilon_{max} = \tau(1 + \nu) / E$  and  $\gamma = \tau / G$

the following relationship can be written

$$G = \frac{E}{2(1 + \nu)}$$

thus  $E$ ,  $G$  and  $\nu$  are not independent properties of a linear elastic material

### 3.7 Transmission of Power by Circular Shafts

the most important use of circular shafts is to transmit mechanical power, such as drive shaft of an automobile, propeller shaft of a ship, axle of bicycle, torsional bar, etc.

a common design problem is the determination of the required size of a shaft so that it will transmit a specified amount of power at a specified speed of revolution without exceeding the allowable stress

consider a motor drive shaft, rotating at angular speed  $\omega$ , it is transmitting a torque  $T$ , the work done is

$$W = T \phi \quad [T \text{ is constant for steady state}]$$



where  $\phi$  is angular rotation in radians, and the power is  $dW / dt$

$$P = \frac{dW}{dt} = T \frac{d\phi}{dt} = T\omega \quad \omega : \text{rad/s}$$

$\therefore \omega = 2\pi f$   $f$  is frequency of revolution  $f: \text{Hz} = \text{s}^{-1}$

$$\therefore P = 2\pi fT$$

denote  $n$  the number of revolution per minute (rpm), then  $n = 60f$

$$\text{thus } P = \frac{2n\pi T}{60} \quad (n = \text{rpm}, T = \text{N-m}, P = \text{W})$$

in U.S. engineering practice, power is often expressed in horsepower (hp),  
 $1 \text{ hp} = 550 \text{ ft-lb/s}$ , thus the horsepower  $H$  being transmitted by a rotating shaft is

$$H = \frac{2n\pi T}{60 \times 550} = \frac{2n\pi T}{33,000} \quad (n = \text{rpm}, T = \text{lb-ft}, H = \text{hp})$$

$$1 \text{ hp} = 550 \text{ lb-ft/s} = 550 \times 4.448 \text{ N} \times 0.305 \text{ m/s} = 746 \text{ N-m/s} \\ = 746 \text{ W} \quad (\text{W} : \text{watt})$$

### Example 3-7

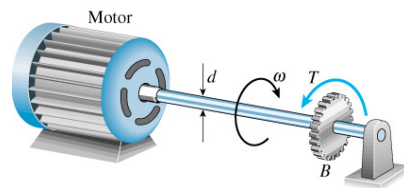
$$P = 30 \text{ kW}, \quad \tau_{all} = 42 \text{ MPa}$$

(a)  $n = 500 \text{ rpm}$ , determine  $d$

(b)  $n = 4000 \text{ rpm}$ , determine  $d$

$$(a) \quad T = \frac{60P}{2\pi n} = \frac{60 \times 30 \text{ kW}}{2\pi \times 500} = 573 \text{ N-m}$$

$$\tau_{max} = \frac{16T}{\pi d^3} \quad d^3 = \frac{16T}{\pi \tau_{all}} = \frac{16 \times 573 \text{ N-m}}{\pi \times 42 \text{ MPa}} = 69.5 \times 10^{-6} \text{ m}^3$$



$$d = 41.1 \text{ mm}$$

$$(b) \quad T = \frac{60 P}{2 \pi n} = \frac{60 \times 30 \text{ kW}}{2 \pi \times 4000} = 71.6 \text{ N-m}$$

$$d^3 = \frac{16 T}{\pi \tau_{all}} = \frac{16 \times 71.6 \text{ N-m}}{\pi \times 42 \text{ MPa}} = 8.68 \times 10^{-6} \text{ m}^3$$

$$d = 20.55 \text{ mm}$$

the higher the speed of rotation, the smaller the required size of the shaft

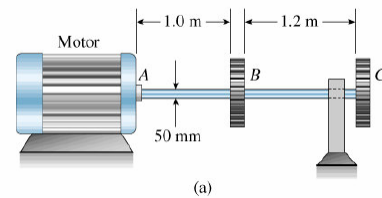
### Example 3-8

a solid steel shaft  $ABC$ ,  $d = 50 \text{ mm}$

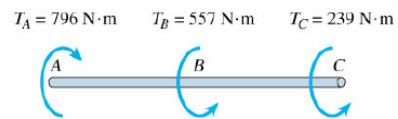
motor  $A$  transmit  $50 \text{ kW}$  at  $10 \text{ Hz}$

$P_B = 35 \text{ kW}$ ,  $P_C = 15 \text{ kW}$

determine  $\tau_{max}$  and  $\phi_{AC}$ ,  $G = 80 \text{ GPa}$



$$T_A = \frac{P_A}{2 \pi f} = \frac{50 \times 10^3}{2 \pi \times 10} = 796 \text{ N-m}$$



similarly  $P_B = 35 \text{ kW}$   $T_B = 557 \text{ N-m}$

$P_C = 15 \text{ kW}$   $T_C = 239 \text{ N-m}$

then  $T_{AB} = 796 \text{ N-m}$   $T_{BC} = 239 \text{ N-m}$

shear stress and angle of twist in segment  $AB$

$$\tau_{AB} = \frac{16 T_{AB}}{\pi d^3} = \frac{16 \times 796}{\pi \times 50^3} = 32.4 \text{ MPa}$$

$$\phi_{AB} = \frac{T_{AB} L_{AB}}{G I_p} = \frac{796 \times 1.0}{80 \times 10^9 \frac{\pi}{32} \times 0.05^4} = 0.0162 \text{ rad}$$

shear stress and angle of twist in segment  $BC$

$$\tau_{BC} = \frac{16 T_{BC}}{\pi d^3} = \frac{16 \times 239}{\pi 50^3} = 9.7 \text{ MPa}$$

$$\phi_{AB} = \frac{T_{BC} L_{BC}}{G I_p} = \frac{239 \times 1.2}{80 \times 10^9 \frac{\pi}{32} 0.05^4} = 0.0058 \text{ rad}$$

$$\therefore \tau_{max} = \tau_{AB} = 32.4 \text{ MPa}$$

$$\phi_{AC} = \phi_{AB} + \phi_{BC} = 0.0162 + 0.0058 = 0.022 \text{ rad} = 1.26^\circ$$

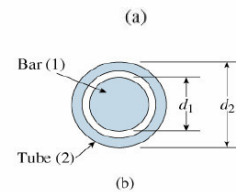
### 3.8 Statically Indeterminate Torsional Members

torsional member may be statically indeterminate if they are constrained by more supports than are required to hold them in static equilibrium, or the torsional member is made by two or more kinds of materials

flexibility and stiffness methods may be used  
only flexibility method is used in the later discussion

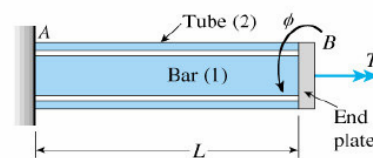


consider a composite bar  $AB$  fixed at  $A$   
the end plate rotates through an angle  $\phi$   
 $T_1$  and  $T_2$  are developed in the  
solid bar and tube, respectively



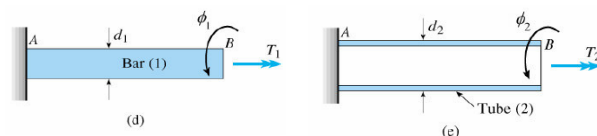
equation of equilibrium

$$T_1 + T_2 = T$$



equation of compatibility

$$\phi_1 = \phi_2$$



torque-displacement relations

$$\phi_1 = \frac{T_1 L}{G_1 I_{p1}} \quad \phi_2 = \frac{T_2 L}{G_2 I_{p2}}$$

then the equation of compatibility becomes

$$\frac{T_1 L}{G_1 I_{p1}} = \frac{T_2 L}{G_2 I_{p2}}$$

now we can solve for  $T_1$  and  $T_2$

$$T_1 = T \left( \frac{G_1 I_{p1}}{G_1 I_{p1} + G_2 I_{p2}} \right) \quad T_2 = T \left( \frac{G_2 I_{p2}}{G_1 I_{p1} + G_2 I_{p2}} \right)$$

and

$$\phi = \frac{TL}{G_1 I_{p1} + G_2 I_{p2}}$$

### Example 3-9

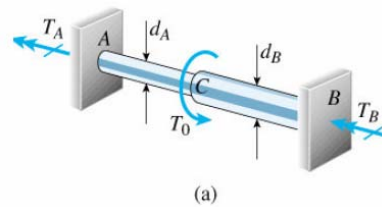
a bar  $ACB$  is fixed at both ends

$T_0$  is applied at point  $C$

$AC$  :  $d_A, L_A, I_{pA}$

$CB$  :  $d_B, L_B, I_{pB}$

determine (a)  $T_A, T_B$  (b)  $\tau_{AC}, \tau_{CB}$  (c)  $\phi_C$



equation of equilibrium

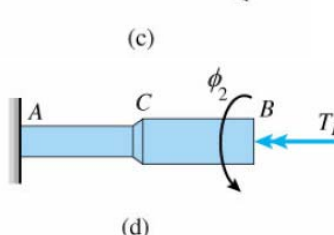
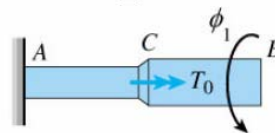
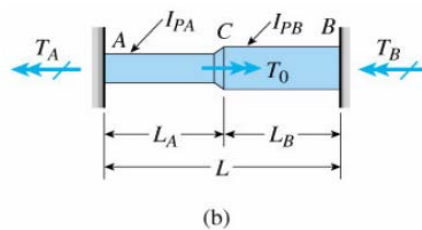
$$T_A + T_B = T_0$$

equation of compatibility

$$\phi_1 + \phi_2 = 0$$

torque-displacement equations

$$\phi_1 = T_0 L_A / G I_{pA}$$



$$\phi_2 = -\frac{T_B L_A}{G I_{pA}} - \frac{T_B L_B}{G I_{pB}}$$

then the equation of compatibility becomes

$$\frac{T_0 L_A}{G I_{pA}} - \frac{T_B L_A}{G I_{pA}} - \frac{T_B L_B}{G I_{pB}} = 0$$

$T_A$  and  $T_B$  can be solved

$$T_A = T_0 \left( \frac{L_B I_{pA}}{L_B I_{pA} + L_A I_{pB}} \right) \quad T_B = T_0 \left( \frac{L_A I_{pB}}{L_B I_{pA} + L_A I_{pB}} \right)$$

if the bar is prismatic,  $I_{pA} = I_{pB} = I_p$

then

$$T_A = \frac{T_0 L_B}{L} \quad T_B = \frac{T_0 L_A}{L}$$

maximum shear stress in  $AC$  and  $BC$  are

$$\tau_{AC} = \frac{T_A d_A}{2 I_{pA}} = \frac{T_0 L_B d_A}{2 (L_B I_{pA} + L_A I_{pB})}$$

$$\tau_{CB} = \frac{T_B d_B}{2 I_{pB}} = \frac{T_0 L_A d_B}{2 (L_B I_{pA} + L_A I_{pB})}$$

angle of rotation at section  $C$  is

$$\phi_C = \frac{T_A L_A}{G I_{pA}} = \frac{T_B L_B}{G I_{pA}} = \frac{T_0 L_A L_B}{G (L_B I_{pA} + L_A I_{pB})}$$

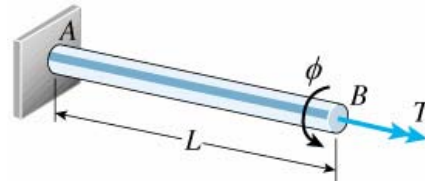
if the bar is prismatic,  $I_{pA} = I_{pB} = I_p$

then

$$\phi_C = \frac{T_0 L_A L_B}{G L I_p}$$

### 3.9 Strain Energy in Torsion and Pure Shear

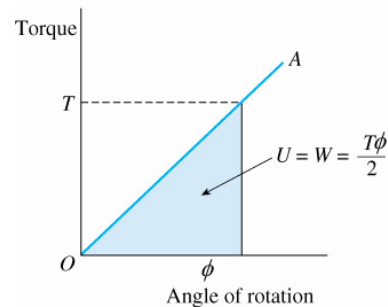
consider a prismatic bar  $AB$  subjected to a torque  $T$ , the bar twists an angle  $\phi$



if the bar material is linear elastic, then the strain energy  $U$  of the bar is

$$U = W = T\phi / 2$$

$$\therefore \phi = TL / GI_p$$



then

$$U = \frac{T^2 L}{2 GI_p} = \frac{GI_p \phi^2}{2 L}$$

if the bar is subjected to nonuniform torsion, then

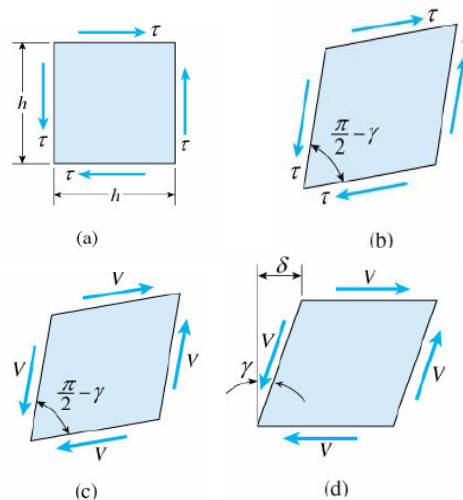
$$U = \sum_{i=1}^n U_i = \sum_{i=1}^n \frac{T_i^2 L_i}{2 G_i I_{pi}}$$

if either the cross section or the torque varies along the axis, then

$$dU = \frac{[T(x)]^2 dx}{2 GI_p(x)} \quad U = \int dU = \int_0^L \frac{[T(x)]^2 dx}{2 GI_p(x)}$$

strain energy density in pure shear

consider a stressed element with each side having length  $h$  and thickness  $t$ , under shear stress  $\tau$  with shear strain  $\gamma$



$\gamma$

the shear force  $V$  is

$$V = \tau h t$$

and the displacement  $\delta$  is

$$\delta = h \gamma$$

for linear elastic material, strain energy stored in this element is

$$U = W = \frac{V \delta}{2} = \frac{\tau \gamma h^2 t}{2}$$

and the strain energy density  $u = U / \text{per unit volume}$ , then

$$u = \tau \gamma / 2 = \tau^2 / 2 G = G \gamma^2 / 2$$

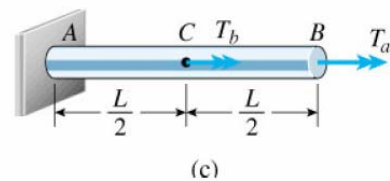
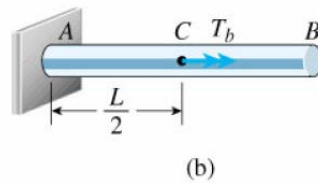
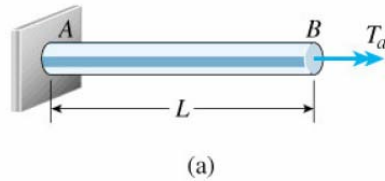
### Example 3-10

a solid circular bar  $AB$  of length  $L$

(a) torque  $T_a$  acting at the free end

(b) torque  $T_b$  acting at the midpoint

(c) both  $T_a$  and  $T_b$  acting simultaneously



$$T_a = 100 \text{ N-m} \quad T_b = 150 \text{ N-m}$$

$$L = 1.6 \text{ m} \quad G = 80 \text{ GPa}$$

$$I_p = 79.52 \times 10^3 \text{ mm}^4$$

determine the strain energy in each case

(a)

$$U_a = \frac{T_a^2 L}{2 G I_p} = \frac{100^2 \times 10^6 \times 1.6 \times 10^3}{2 \times 80 \times 10^3 \times 79.52 \times 10^3} = 1.26 \text{ J (N-m)}$$

(b)

$$U_b = \frac{T_b^2 (L/2)}{2 G I_p} = \frac{T_b^2 L}{4 G I_p} = 2.83 \text{ J}$$

$$\begin{aligned}
 (c) \quad U_c &= \sum_{i=1}^n \frac{T_i^2 L_i}{2 G_i I_{pi}} = \frac{T_a^2 (L/2)}{2 G I_p} + \frac{(T_a + T_b)^2 (L/2)}{2 G I_p} \\
 &= \frac{T_a^2 L}{2 G I_p} + \frac{T_a T_b L}{2 G I_p} + \frac{T_b^2 L}{4 G I_p} \\
 &= 1.26 \text{ J} + 1.89 \text{ J} + 2.83 \text{ J} = 5.98 \text{ J}
 \end{aligned}$$

Note that (c) is not equal to (a) + (b), because  $U \sim T^2$

### Example 3-11

a prismatic bar  $AB$  is loaded by a distributed torque of constant intensity  $t$  per unit distance

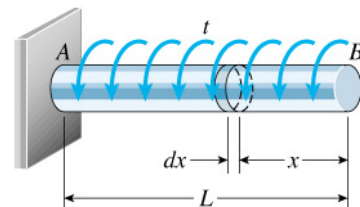
$$t = 480 \text{ lb-in/in} \quad L = 12 \text{ ft}$$

$$G = 11.5 \times 10^6 \text{ psi} \quad I_p = 18.17 \text{ in}^4$$

determine the strain energy

$$T(x) = tx$$

$$\begin{aligned}
 U &= \int_0^L \frac{[(tx)]^2 dx}{2 G I_p} = \frac{1}{2 G I_p} \int_0^L (tx)^2 dx = \frac{t^2 L^3}{6 G I_p} \\
 &= \frac{480^2 \times (12 \times 12)^3}{6 \times 11.5 \times 10^6 \times 17.18} = 580 \text{ in-lb}
 \end{aligned}$$

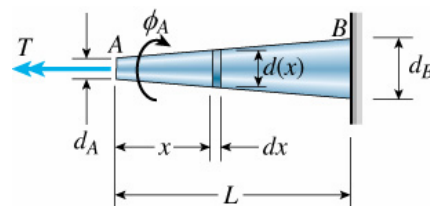


### Example 3-12

a tapered bar  $AB$  of solid circular cross section is supported a torque  $T$

$$d = d_A \sim d_B \text{ from left to right}$$

determine  $\phi_A$  by energy method





$$W = \frac{T \phi_A}{2}$$

$$I_p(x) = \frac{\pi}{32} [d(x)]^4 = \frac{\pi}{32} \left( d_A + \frac{d_B - d_A}{L} x \right)^4$$

$$U = \int_0^L \frac{[T(x)]^2 dx}{2 G I_p(x)} = \frac{16 T^2}{\pi G} \int_0^L \frac{dx}{\left( d_A + \frac{d_B - d_A}{L} x \right)^4}$$

$$= \frac{16 T^2 L}{3 \pi G (d_B - d_A)} \left( \frac{1}{d_A^3} - \frac{1}{d_B^3} \right)$$

with  $U = W$ , then  $\phi_A$  can be obtained

$$\phi_A = \frac{32 T L}{3 \pi G (d_B - d_A)} \left( \frac{1}{d_A^3} - \frac{1}{d_B^3} \right)$$

same result as in example 3-5

### 3-10 Thin-Walled Tubes

### 3-11 Stress Concentrations in Torsion

### 3-12 Nonlinear Torsion of Circular Bars